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Plasma-based space radiation mimicking for space radiobiology and electronics testing

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> Strathclyde Centre for Doctoral Training P-PALS Plasma-based Particle and Light Sources

> > Strathclyde Space Institute

& The Cockcroft Institute



Radiation is a fundamental driver of knowledge.



Greek Philosophy: Allegory of the cave; Analogy of the sun Plato, Politeia, 380 BC

Radiation is a fundamental driver of knowledge.



Greek Philosophy: Allegory of the cave; Analogy of the sun Plato, Politeia, 380 BC The Sun: fusion and plasma processes send broadband **photon** and plasma **particle** radiation to Earth



Atmosphere protects us from too intense and too hard **photon** flux

Earth provides the right amount of protection: too much photons or particles incident on Earth would prevent life to occur, but a little amount is required for genetic evolution



Magnetosphere protects us from too intense charged **particle** flux (electrons, protons, ions..)

Aurora Borealis – Northern lights (or for the Southerners Aurora Australis)



Electron acceleration in the outer radiation belt

Ionization effects from electrons entering Earth at the magnetic poles

Hidding / University of Strathclyde & SCAPA: Radiation Hardness Assurance

No protection in space – radiation major obstacle for space exploration

- In space, Earth's protection via the magnetic field and atmosphere is lost
- Space radiation can be extremely versatile (electrons, protons, ions, neutrons, photons)
- Space radiation can kill satellites/missions/astronauts
- Testing & selection of space-grade electronics is one of the most money- and timeconsuming factors in spacecraft design and operation. Up to 1/3 of total mission costs can be consumed by radiation hardness assurance (RHA)
- Each electronic component batch must be tested/certified via standardized method major cost driver! ESA: satellite market 80 G€/a
- RHA of space electronics can be similarly complex as cancer radiotherapy. Multiple tests, with different types of beams at different facilities may be required
- Performance/size/weight of electronics used in space lags behind mass production COTS by several generations
- Space exploration is a vibrant and expanding field of interest with large governmental and industrial impact
- In the EU alone 12 billion euros are being invested between 2014 & 2020 to further Europe's presence in space





Various damage effects e.g. on electronics

- Total ionizing dose (TID), cumulative damage
- Single Event Effects (SEE)
- Surface charging (low energy electrons/protons)
- Deep Dielectric Discharge (DDD)
- ..



Components separated by dielectrics, protective layers of passivating insulators and glass.

Space radiation can bridge isolation between components, or generate fields/charge within components.

Figures from Aerospace Corporation Magazine

NMOS: gate allows current to flow above threshold voltage SiO2 gate oxide should be ideal insulator, BUT is ionized by received dose Electron/hole pairs are created in SiO2, electrons drift away, but fraction of holes are trapped and accumulate.

Large positive charge has same effect as positive voltage applied to gate: NMOS spuriously turns on, remains on.

PMOS analogoulsy: When radiation has produced enough positive charge in gate oxide, device stays off permanently.

In CMOS logical circuit: output will be frozen at "0" or "1"

Hardened gate oxides trap much less holes than commercial mass products (material sciences)

Adjacent transistors are separated by thick field oxide layers, where enough positive charge can be trapped to connect both transistors etc.

This "edge leakage" is today often the dominant, and limiting total-dose effect: transistors collectively leak too much for power supply





Displacement damage

PKA: Primary Knock on Atom

Displacement damage energy thresholds in Si: $E_d \sim 25 \text{ eV}$ (single lattice atom, Frenkel pair)

Neutrons: $E_n > 185 \text{ eV}$ Electrons: $E_e > 255 \text{ keV}$

Energy transfer in $T_{\rm max} = \frac{2ME(E+2mc^2)}{(m+M)^2c^2+2ME}$ (relativistic)

 $T_{\rm max} = E \frac{4Mm}{(m+M)^2}$ (nonrelativistic)

Disruption of crystalline semiconductor lattice structure leads to degradation of electric performance NIEL: non-ionizing energy loss (about 0.1% of total energy loss) DDD: displacement damage dose



Dale et al.

-.lun

100

≭– Akkerman et al

ASTM E 722

Summers et al

Protons

Neutrons

Electrons

-Huhtinen and Arnio

Uasilescu & Lindstroem

Si



Electrons

10

Particle Energy (MeV)

Protons

Neutrons

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100

10

0.1

+ 0.01 0.1

NIEL, (keVcm2/g)

1000

Single Event Effects

High energetic proton or ion generates ionization track

Number of charge pairs propotional to LET: linear energy transfer (in MEV-cm²/mg) Stopping power, Bragg peak

e.g., in NMOS: short is generated between substrate (grounded) and drain: above critical charge, spike current may generate single-event upset (SEU)



ESA Herschel, 2009:

-SEU in RAM of the Local Oscillator Control Unit (LCU) of HIFI telescope activated an emergency switch off. -This switch was designed to protect the local oscillators against damage from a drop in spacecraft power supply (28 V).

- But now the switch was activated while power supply was still up, resulting in an overvoltage spike.
- overload in one of the power converters, leading to permanent failure of a diode.

=> months downtime

Example: killer electrons

- Early example: "Anik Panic" 1994:
- Control over Canadian Anik Satellites lost after
- Bombardment with radiation (electrons)
- Killer electrons usually occur most strongly in outer van Allen Belt, distance to Earth 3-9 Earth radii
- E.g., GPS /Galileo satellites at approx. 22000 km, MEO (Medium Earth Orbit) is passed by every spacecraft (manned or unmanned) going beyond LEO



Image Credit: L. J. Lanzerotti, Bell Laboratories, Lucent Terchnologies, Inc.

- Telephone/cell phone/radio/television/navigation can be heavily affected, killer electrons can knock out computers, degrade solar arrays, pierce spacesuits, damage tissues of astronauts, endanger Mars missions etc.
- In addition: Solar activity can push radiation belts much closer to Earth!
- E.g., "Halloween Storm" 2003: SAMPEX (Solar Anomalous and Magnetospheric Particle Explorer) detected: center of outer van Allen belt as close as 6 miles to Earth!
- ⇒ 30 satellites reported malfunctions, one was a total loss (avg. total satellite costs ~500 M€)

Information on the space radiation effects can be obtained using the following three ways:

- on-board experiments in space; 1.
- ground-based radiation testing;
- mathematical modeling. 3.

The space experiments may provide the most reliable information. However, it is difficult to obtain the necessary full database both on the space radiation features and on the appearing effects in spacecraft materials and devices in such experiments. Besides, space experiments are very expensive. So the main amount of information is obtained using the ground-based radiation testing of the spacecraft materials and devices, as far as the mathematical modeling.

In turn, two approaches are possible in the ground-based testing:

1. reproduction of the space radiation features (the ion composition, energy spectra, angular distribution, intensity) in laboratory equipment as exactly as possible;

- 2. radiation testing under the following simplifications:
 - usage of monoenergetic electron critical state-of-the-art approach distributed energy spectra. with change of the charged particle fluxe
 increase to the radiation intensition by a factor of the charged particle fluxe

 - increase to the radiation intensities by a factor of 10²-10³ compared to conditions in space for reduction of the tasting time.

Evidently, the first approach specified above is very difficult to realize technically, and the second approach requires enough serious scientific ground, to the led knowledge of physical mechanisms of the radiation end and accelerators to the

rescue!

Space Radiation is a complex mix of electrons, protons/ions, neutrons and **broadband**, typically with exponential / power–law reduction of flux towards higher particle energies:



Fig. 5. Energy spectra of particles inside the module "Columbus" formed by (a) ERB and (b) GCR protons: (1) primary particles and (2) and (3) neutrons and protons inside the module.

Linacs and cyclotrons inherently produce **monoenergetic**, "unnatural" beams. Reproduction of the exponential/power-law shaped spectral flux would be desirable



Fig. 5. Energy spectra of particles inside the module "Columbus" formed by (a) ERB and (b) GCR protons: (1) primary particles and (2) and (3) neutrons and protons inside the module.

Spectral flux in space vs. linac/cyclotron output



- Spectra are substantially different, even diametrally opposed.
- Since charge/dose deposition and resulting damaging is fundamentally different, conventional approaches are insufficient

Spectral flux & TID when passing through (AI) shielding:



Königstein, Karger et al., Journal of Plasma Physics, 2012

Various kinds of damage (SEU, DDD, IESD..)

For example, internal electrostatic discharge (IESD): space "killer" electrons are accumulated in dielectrics due to low conductivity. E-field builds up and if it exceeds breakdown threshold of the dielectric \Rightarrow discharge \Rightarrow damages of surrounding electronics \Rightarrow spacecraft failure



Directionality and energy distribution of electron flux matters

Van Allen belt acceleration mechanisms and killer electrons

Acceleration mechanisms in space are an own vibrant field of research



Electron acceleration in the outer radiation belt

Horne et al., "Wave acceleration of electrons in the van Allen radiation belts", Nature 437, 2005 Chen et al., "The energization of relativistic electrons in the outer van Allen radiation belt", Nature Physics 3, 2007 Horne et al., "Plasma astrophysics: Acceleration of **killer electrons**", Nature Phys. 3, 2007 Horne et al., "Gyro-resonant electron acceleration at Jupiter", Nature Physics 4, 2008

Van Allen belt acceleration mechanisms and killer electrons



Electron acceleration in the outer radiation belt

Inner Belt: 1,000-6,000 km Lower boundary can extend down to 200 km (ISS at 400 km) depending on solar activity and the South Atlantic Anomaly (SAA)

Dominated by protons

Outer Belt: 13,000-60,000 km

Dominated by electrons

Use plasma accelerators to reproduce space radiation for RHA

- DE Patent (2010) and US/PCT patents (2011/12) (RadiaBeam & UCLA US United States Patent 8947115, 2015)
- ESA-funded seed activities: ESA NPI "Study of Space Radiation Effects with Laser-Plasma-Accelerators", 2011-2013, ESA GSP "Laser-Plasma-Accelerator's Potential to Radically Transform Space Radiation Testing", 2012-2014

(12) United States Patent Rosenzweig et al.			(10) Patent No.: US 8,947,115 B2 (45) Date of Patent: Feb. 3, 2015			
(54)	METHOI COMPON	METHOD OF TESTING ELECTRONIC COMPONENTS		USPC		
(75)	Inventors:	James Rosenzweig, Los Angeles, CA (US); Alex Y. Murokh, Encino, CA (US); Bernhard Hidding, Dusseldorf	(56)	(56) References Cited		
		(DE)		U.S. PATENT	DOCUMENTS	
(73)	Assignee:	Radiabeam Technologies, LLC, Santa Monica, CA (US)		5,179,279 A * 1/1993 6,476,400 B1 * 11/2002	Millard et al	

How to reproduce exponential/power law space radiation flux mit plasma accelerators?

Various options, e.g.

- Ti:Sapphire plasma acceleration (with solids or with underdense targets)
- CO2 laser plasma acceleration
- Linac PWFA based

electronic component



Space radiation from laser-solid interaction (TNSA-style)

Radiation belt spectral flux calculations

Using NASA AE8/AE9 & AP9 models: Electrons > 100 keV



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3. Setup, adapt testing techniques, monitor flux and irradiate devices



Results: first accurate reproduction of space radiation, also production of broadband protons, significant degradation of optocoupler performance

Main diagnostics: Image plate stack for dose monitoring + magnet spectrometer on axis



Metallic target foil after laser shots







Ti:Sapphire: Proof-of-concept runs w/ 150 TW Arcturus laser at University Düsseldorf (electrons mainly, via laser-solid interaction) and at VULCAN PW (protons)





Feb. 3, 2015

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US 8,947,115 B2

FIG. 4B



other electronics with Strathclyde per gamma rays



Proof-of-concept studies and experimental results

Degradation of optocoupler performance: Current Transfer Ratio (CTR)



Laser-plasma-based Space Radiation Reproduction in the Laboratory, Sci. Reports 7: 42354 (2017)

Hidding / University of Strathclyde & SCAPA: Radiation Hardness Assurance

Space radiation from PWFA-style systems







Tail decelerates

.



- Good fit for killer electrons in radiation belts of Earth
- E.g. linacs such as CLARA/UK, SPARC/Italy, CTF3/CERN...

Nature is producing the exponential flux similar as we do in the lab!

volcanic mass ejection



"the gas jet"

sunlight and impact ionization produces Jovian aurora



"the laser/linac plasma source"

then the gyromagnetic interaction in the Jovian magnetic fields accelerates the electrons..

Linac – PWFA driven; Jovian missions (ESA JUICE, NASA)



- Jovian electron flux is orders of magnitude more intense than in Earth radiation belts
- Jovian proton flux ~ 2 orders of magnitude less than electrons

Space rad designer flux: combine different exponential spectra



Figure 6. Spectral flux in Jupiter's radiation belt. (a) Flux at a distance of 5 R_J and 9 R_J , (b) overlaying three exponential spectra can reproduce with high accuracy the flux expected at 9 R_J .

General considerations:

- electron energies ~order of magnitude lower than proton energies e.g. in Earth rad belts
- electron flux e.g. in Earth rad belts exceeds proton flux by orders of magnitude
- rad. damage by protons/ions much higher than by electrons;



Protons of few MeV energies are getting increasingly important especially for sub-45 nm technology (secondaries) Extremely high flux on target surface, when close to the target Then massive flux reduction because of a) divergence of beam, b) time-of-flight differences



Figure 5. Reduction of exponential-energy electron flux due to energy-dependent velocities and divergence. In (a), the flux of a beam with $T_{\text{eff}} = 0.35 \text{ MeV}$, Q = 100 nC, and a divergence $\theta = 25^{\circ}$ through a DUT area of 1 cm^2 is calculated at distances 0.1, 1, and 10 cm behind target (note the logarithmic scaling). Next, the influence of the divergence is visualized by plotting the flux through 1 cm^2 after a distance 1 cm (b) and 10 cm (c) for the beam with parameters as in (a), but for a hypothetical divergence of $\theta = 0^{\circ}$ and 25° .

Space radiation from CO2-laser systems

Reproducible ion acceleration with 10 µm CO₂

- At BNL ATF, observed production of MeV proton and helium beams via collisionless shock acceleration
- Optical shaping of targets using controlled pre-pulse provides means to reproducible ion beams

Recently Published: O. Tresca, N. P. Dover, N. Cook, C. Maharjan, M. N. Polyanskiy, Z. Najmudin, P. Shkolnikov, and I. Pogorelsky. Phys. Rev. Lett. **115**, 094802 (2015).

1.1 0.9

0.7

0.1

Imperial College

1.1

() 0.5 Ш^{аа} 0.3 x10⁸ ∎1.4

1.2

0.8

Stony Brook University

1.5

1.3

Main pulse a_0



Efficient scaling of acceleration with laser intensity

- Collisionless shock acceleration exhibits strong scaling with laser energy, E_{ion} ~ I_L/n_e
 - Peak energies exceed those predicted by hole-boring RPA models
- ATF-II projects 100 TW peak intensities, a₀ ≈ 10 - protons energies > 140 MeV at full power are possible!
 - Variation with peak target density, thickness, and laser parameters requires further consideration

Work supported by U.S. DOE Contract No. DE-AC02-98CH10886, U.S. DOE Grant No. DE-FG02-07ER41488, UK EPSRC Grant No. EP/K022415/1, and BNL/LDRD Grant No. 12-032.

Imperial College John Adoms Institute for Accelerator Science







Simulations performed using EPOCH in 2D. Images courtesy N. Cook and O. Tresca.



Potential and future development of the scheme

Potential and future development of the scheme

Irradiation times, assuming daily fluence on Nav-orbit: 3 x 10¹² cm⁻²:

- 6.5 hrs w/ LINAC @ 1.3 x 10⁸ cm⁻²s⁻¹
- 3.9 hrs w/ laser-plasma-accelerator @ 2.1×10^7 cm⁻² per shot at 10 Hz rep rate (today's standard)
- 140 sec w/ laser-plasma-accelerator @ 2.1 x 10⁷ cm⁻² per shot at 1 kHz rep rate (avantgarde but existing and already used laser systems, e.g. Schmid et al., PRL 2009). Requires tape drive targets / droplets / gas jets.
- 1.4 sec w/ laser-plasma-accelerator @ 2.1 x 10⁷ cm⁻² per shot at 100 kHz rep rate (appearing on the horizon, especially efficient fiber and thin disc lasers. Remark: Theoretical value, limiting factors: vacuum system & too high peak flux)

Using the full solid angle of radiation for testing:

Laser development European Anti-



Options for (space) radiobiology



Radiobiology: VHEE (very high energy electrons) dosimetry, D. Jaroszynski et al Phys. Med. Biol.





dose mapping measurement vs. ca LWFA-generated electrons in water phantom



vs. calculation for 135 MeV



Düsseldorf w/ kHz laser:

Xanthoria elegans (Flechte)



control sample CLSM photos (confocal laser scanning microscope)



sample after 360 sec.



sample after 480 sec. I. Halezki, F. Gaußmann et al.





Potential for dedicated RHA beamline

Scottish Centre for the Application of Plasma-based Accelerators

- Collaborative research opportunity for Glasgow & Scotland, the UK and beyond
- ~£10M investment + additional infrastructure funds (SFC, SUPA, UoS..)
- Accelerator and Light Source R&D
- Strong engagement in European and other large projects
- In-depth programme of applications, knowledge exchange & commercialization



- 3 high-power laser systems, initially up to 350 TW
- 3 shielded radiation caves, fully vibration-isolated, w/ 2000 tons of concrete shielding
- up to 7 accelerator application beam lines
- ~1200 m² on two levels
- High-energy particle beams: electrons, protons, ions, positrons, neutrons
- High-energy photon beams: fs duration, (coherent) VUV, X-ray & gamma-rays

Summary & Outlook

- Plasma accelerators inherent ability to produce broadband, "exponential" beams is a highly desirable feature for space rad reproduction & testing
- Ti:Sapphire, CO2 and linac driven plasma acceleration useful
- High fluence is desirable (test over full life cycle of satellite)?
- Laser development (especially high rep rate) desirable
- Ti:Sapphire: over- and underdense interaction possible, both electrons and protons (e.g. TNSA). In laser-solid interaction, nC per shot possible, but rep rate limited. Near-future few-TW Ti:Sapphire lasers and underdense interaction may allow for kHz output at Earth radiation belt scale energies (up to 10 MeV). Jovian energy levels (up to 150 MeV) accessible with 10-100 TW lasers.
- Linac: electrons (both Earth and Jovian). Longer term towards application: High rep rates possible, nC per shot & large fluence
- Develop test standard together with National Physical Laboratory, CLF and Cl
- Further tests with active electronics
- Space radiobiology tests
- Develop European R&D programme jointly with ESA and other partners (e.g. H2020)

Literature

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- Design considerations for the use of laser-plasma accelerators for advanced space radiation studies, T Königstein, O. Karger, G. Pretzler, J. B. Rosenzweig, B. Hidding, Journal of Plasma Physics, Volume 78 / Special Issue 04 / August 2012, pp 383-391
- ESA NPI project "Study of Space Radiation Effects with Laser-Plasma-Accelerators" final report, 2014
- ESA GSP project "Laser-Plasma-Accelerator's Potential to Radically Transform Space Radiation Testing", 2014
- Proof-of-concept Ti:Sapphire laser experiment, Sci. Reports 2017